# Modeling and calculations of rarefied gas flows: DSMC vs kinetic equation

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51 IUVSTA Workshop Modern Problems & Capability of Vacuum Gas Dynamics Värmdö, July 9-12, 2007

to pump a gas

- to pump a gas
- to maintain a low pressure

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Gas dynamics is a basis of vacuum technology

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In vacuum systems

$$10^5 \, \mathrm{Pa} > \mathrm{pressure}$$

$$> 10^{-9} \text{ Pa}$$

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In vacuum systems

$$10^5~{
m Pa}~>~{
m pressure}~>10^{-9}~{
m Pa}$$
  $10^{-8}~{
m m}~<~{
m mean~free~path}<~10^6~{
m m}$ 

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m Pa}$$
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m Pa}$   $10^{-8}~{
m m}$  < mean free path <  $10^6~{
m m}$   $10^{-8}$  < Kn  $< 10^9$ 

Free molecular regime

 $\mathsf{Kn}\gg 1$ 

Free molecular regime

 $Kn \gg 1$ 

Every particle moves independently on each other.

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Test particle Monte Carlo method

Free molecular regime

 $Kn \gg 1$ 

Every particle moves independently on each other.

- Test particle Monte Carlo method
- Method of angle elements

Hydrodynamic regime  ${\rm Kn} \ll 1$ 

Hydrodynamic regime

 $\mathrm{Kn}\ll 1$ 

Continuum mechanics equations are solved

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 $Kn \ll 1$ 

Continuum mechanics equations are solved

The methods are well developed and well known.

Hydrodynamic regime

 $Kn \ll 1$ 

Continuum mechanics equations are solved

- The methods are well developed and well known.
- There are many commercial codes.

Transition regime  ${\rm Kn} \sim 1$ 

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Navier-Stokes eq. is not valid Intermolecular collision cannot be neglected

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Direct simulation Monte Carlo method is applied

Transition regime

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Navier-Stokes eq. is not valid Intermolecular collision cannot be neglected

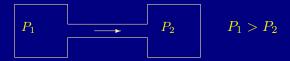
- Direct simulation Monte Carlo method is applied
- Kinetic Boltzmann equation is solved

# TYPICAL PROBLEMS

#### Poiseuille flows



#### Poiseuille flows

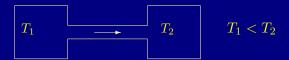


To be calculated:

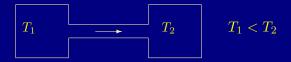
 $\dot{M}$  mass flow rate

density (or pressure) distribution

# Non-isothermal flows, thermal creep



## Non-isothermal flows, thermal creep



To be calculated:

 $\dot{M}$  mass flow rate

Q heat flow rate

density (or pressure) distribution

# Thermomolecular pressure difference



 $\dot{M}=0$  no mass flow

# Thermomolecular pressure difference



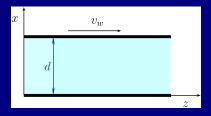
 $\dot{M}=0$  no mass flow

To be calculated:

What is the pressure ratio

$$\frac{P_2}{P_1} = \left(\frac{T_2}{T_1}\right)^{\gamma} \qquad 0 \ge \gamma \ge 0.5$$

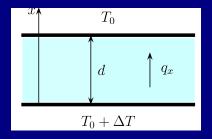
#### Couette flow



To be calculated:

 $P_{xz}$  shear stress

## Heat transfer between two plates

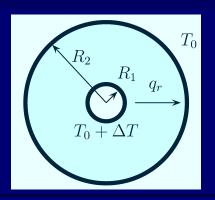


To be calculated:

 $q_x$  Heat flux

## Heat transfer between two cylinders

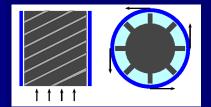
#### Pirani sensor



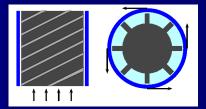
To be calculated:

 $q_r$  Heat flux

# Holweck pump



## Holweck pump

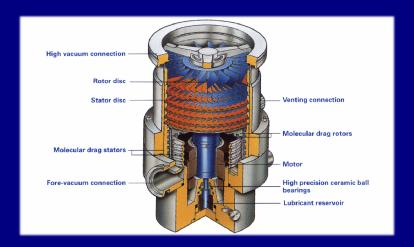


#### To be calculated:

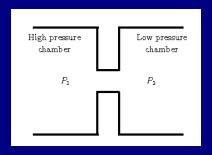
- Compression ratio
- Pumping speed
- Torque



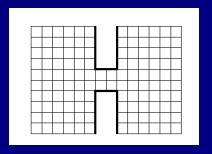
## Combination Holweck and turbomolecular pumps



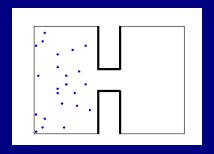
# Direct Simulation Monte Carlo method DSMC



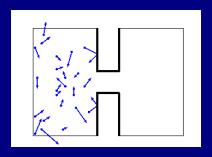
Gas flow through a short tube.



Flow region is divided into cells

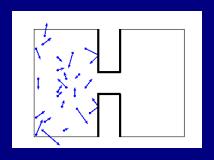


M model particles are considered. Their positions  $r_i$  and velocities  $v_i$  are saved.

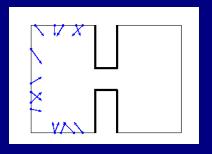


Time is advanced in steps  $\Delta t$ . New positions are calculated

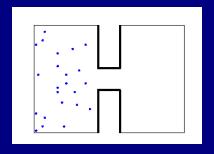
$$r_{i,new} = r_{i,old} + v_i \Delta t$$



Gas-surface interaction is simulated. Some particles are removed.

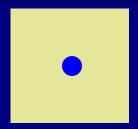


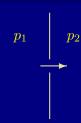
New particles are generated.

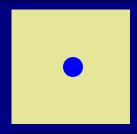


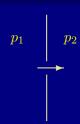
Intermolecular interactions are simulated. Macroscopic quantities are calculated.

All steps are repeated many times in order to reduce the statistical noise.



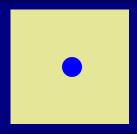


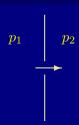




Reduced flow rate

$$W = \frac{\dot{M}}{\dot{M}_0}, \qquad \dot{M}_0 = \frac{\sqrt{\pi}a^2}{v_m}p_1$$





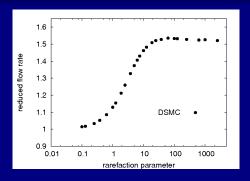
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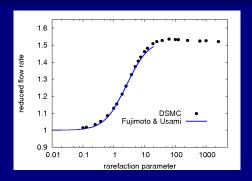
#### Rarefaction parameter

$$\delta = rac{PR}{\mu v_m} \propto rac{1}{\mathsf{Kn}}, \qquad v_m = \sqrt{rac{2R_gT}{M}}$$



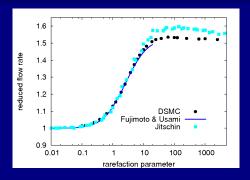


Sharipov, AIAA Journal (2002); J. Fluid Mech. (2004)



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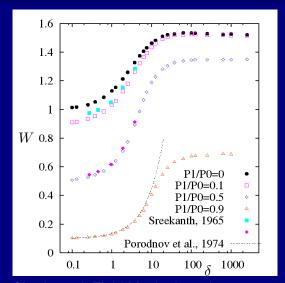
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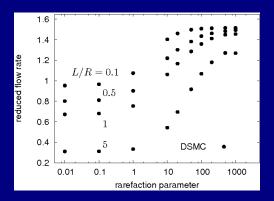
# DSMC, Orifice flow into background gas



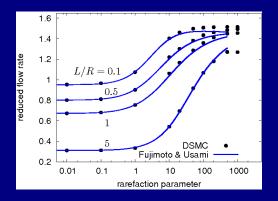
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# DSMC, Flow into vacuum through a short tube



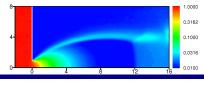
# DSMC, Flow into vacuum through a short tube



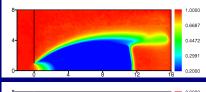
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# DSMC, Orifice flow into background gas

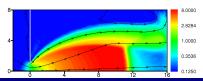
Flow-field at  $p_2/p_1 = 100$  and  $\delta = 1000$ 



 $\varrho/\varrho_0$  density



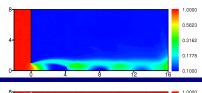
 $T/T_0$  temperature



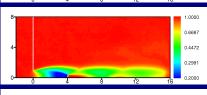
Local Mach number

# DSMC, Orifice flow into background gas

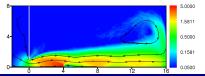
Flow-field at  $p_2/p_1 = 10$  and  $\delta = 1000$ 



 $arrho/arrho_0$  density



 $T/T_0$  temperature



Mach number



#### Advantages

The idea is very clear

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- It is easy to simulate non-elastic collisions occurring in polyatomic gases
- Even more complicated phenomena like dissociation, ionization etc. are considered without effort.
- The books by G.A. Bird contain numerical codes that can be modified and used in engineer calculations.

# NICE!



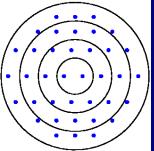
#### IS IT UNIVERSAL REMEDY?



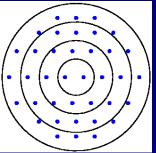
#### **UNFORTUNATELY NOT**



#### Axisymmetrical flows:

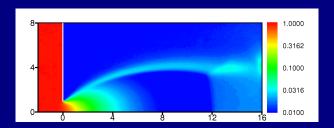


#### Axisymmetrical flows:

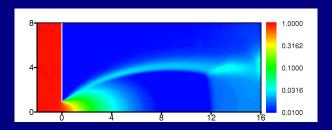


It is necessary to use the radial weighting factor.

#### Flow with high variation of density



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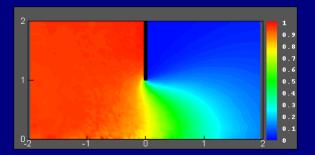


It is necessary to use the longitudinal weighting factor.

# DSMC, Statistical noise

Kn=0.01 and  $P_2/P_1 = 0$ 

#### Density distribution



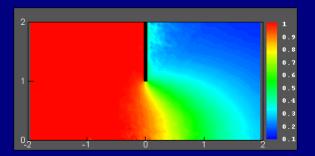
Number of samples  $10^4$  Calculation time - few hours



# DSMC, Statistical noise

Kn=0.01 and  $P_2/P_1 = 0$ 

#### Temperature distribution

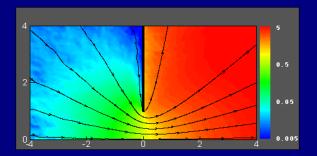


Number of samples  $10^4$  Calculation time - few hours



Kn=0.01 and  $P_2/P_1 = 0$ 

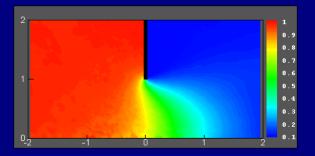
#### Local Ma distribution





Kn=0.01 and  $P_2/P_1 = 0.1$ 

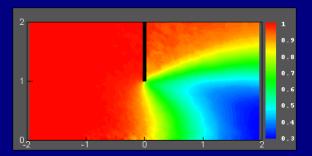
### Density distribution





Kn=0.01 and  $P_2/P_1 = 0.1$ 

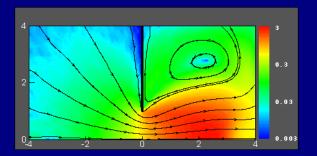
### Temperature distribution





Kn=0.01 and  $P_2/P_1 = 0.1$ 

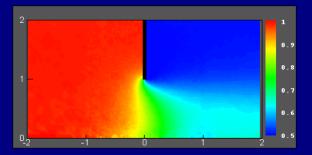
#### Local Ma distribution





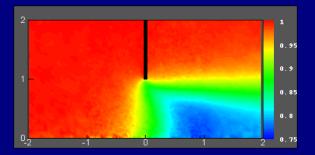
Kn=0.01 and  $P_2/P_1 = 0.5$ 

### Density distribution



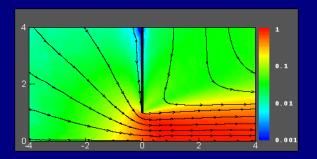
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### Temperature distribution



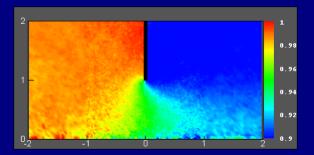
Kn=0.01 and  $P_2/P_1 = 0.5$ 

#### Local Ma distribution



Kn=0.01 and  $P_2/P_1 = 0.9$ 

### Density distribution

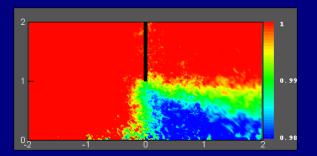


Number of samples 10<sup>6</sup>
Calculation time - few weeks



Kn=0.01 and  $P_2/P_1 = 0.9$ 

#### Temperature distribution

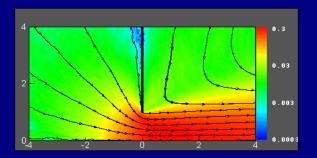


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Kn=0.01 and  $P_2/P_1 = 0.9$ 

#### Local Ma distribution



Number of samples 10<sup>6</sup>
Calculation time - few weeks



Statistical noise is very significant at low Mach number

disadvantages

### disadvantages

A large computer memory

#### disadvantages

- A large computer memory
- Significant non-uniformity of model particle distribution

#### disadvantages

- A large computer memory
- Significant non-uniformity of model particle distribution
- Significant statistical noise

# Kinetic equation

# Velocity distribution function

 $\overline{V} = v - u$ 

$$f(t, m{r}, m{v}) dm{r} dm{v}$$
 number of molecules in  $dm{r} dm{v}$   $n(t, m{r}) = \int f(t, m{r}, m{v}) dm{v}$  - number density  $m{u}(t, m{r}) = rac{1}{n} \int m{v} f(t, m{r}, m{v}) dm{v}$  - bulk velocity  $P(t, m{r}) = rac{m}{3} \int V^2 f(t, m{r}, m{v}) dm{v}$  - pressure  $T(t, m{r}) = rac{m}{3nk} \int V^2 f(t, m{r}, m{v}) dm{v}$  - temperature  $m{q}(t, m{r}) = rac{m}{2} \int V^2 m{V} f(t, m{r}, m{v}) dm{v}$  - heat flux vector



# Boltzmann equation

$$\begin{split} \frac{\partial f}{\partial t} + \boldsymbol{v} \cdot \frac{\partial f}{\partial \boldsymbol{r}} &= Q(ff_*) \\ Q(ff_*) &= \int \left( f'f_*' - ff_* \right) |\boldsymbol{v} - \boldsymbol{v}_*| b \mathrm{d}b \, \mathrm{d}\varepsilon \, \mathrm{d}\boldsymbol{v}_* \end{split}$$

 $oldsymbol{v}'$  and  $oldsymbol{v_*}'$  - pre-collision molecular velocities

 $oldsymbol{v}$  and  $oldsymbol{v}_*$  - post-collision molecular velocities

# Boltzmann equation

Discrete velocity method:

$$\boldsymbol{v}_1, \, \boldsymbol{v}_2, \, \dots, \boldsymbol{v}_N,$$

The BE is split into N differential eqs. coupled via the collisions integral



# Model equations

The collision integral is simplified

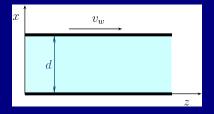
**BGK** model

$$Q(ff_*) = \nu \left( f^M - f \right)$$

S model

$$Q(ff_*) = \nu \left\{ f^M \left[ 1 + \frac{2m(\boldsymbol{q} \cdot \boldsymbol{V})}{15n(kT)^2} \left( \frac{mV^2}{2kT} - \frac{5}{2} \right) \right] - f \right\}$$





 $P_{xz}$  shear stress?

#### Input equation

$$c \frac{\partial \phi}{\partial x} = \delta(u - \phi), \quad u = \frac{1}{\sqrt{\pi}} \int \mathbf{e}^{-c^2} \phi(x, c) dc$$
 
$$\delta = \frac{Pd}{\mu v_m} \propto \frac{1}{\mathsf{Kn}}$$

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Free-molecular regime,  $\delta = 0$  analytical solution

$$P_{xz}^{fm} = \frac{p}{\sqrt{\pi}} \frac{v_w}{v_m}, \quad v_m = \sqrt{2kT/m}$$

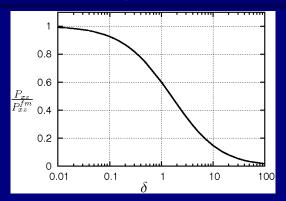


Transitional regime,  $\delta \sim 1$ 

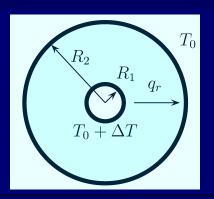
Equation is solved numerically in few seconds

Transitional regime,  $\delta \sim 1$ 

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#### Pirani sensor



To be calculated:

 $q_r$  Heat flux



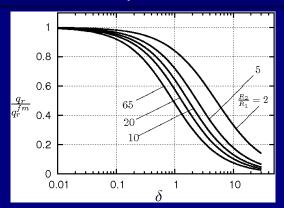
#### Input equation

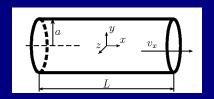
$$\begin{split} c_r \frac{\partial h}{\partial r} - \frac{c_\varphi}{r} \frac{\partial h}{\partial \theta} &= \delta \left[ \upsilon + \tau \left( c^2 - \frac{3}{2} \right) + \frac{4}{15} q c_r \left( c^2 - \frac{5}{2} \right) - h \right], \\ \upsilon(r) &= \frac{1}{\pi^{3/2}} \int \exp(-c^2) h(r, \boldsymbol{c}) \, \mathrm{d}\boldsymbol{c}, \\ \tau(r) &= \frac{1}{\pi^{3/2}} \int \exp(-c^2) h(r, \boldsymbol{c}) \left( \frac{2}{3} c^2 - 1 \right) \, \mathrm{d}\boldsymbol{c}, \\ q(r) &= \frac{1}{\pi^{3/2}} \int \exp(-c^2) h(r, \boldsymbol{c}) \left( c^2 - \frac{5}{2} \right) c_r \, \mathrm{d}\boldsymbol{c}. \end{split}$$

Free-molecular regime,  $\delta = 0$  analytical solution

$$q_r^{fm}(r) = \frac{p \, v_m \, R_1}{\sqrt{\pi} r} \frac{\Delta T}{T_0},$$

Transitional regime, $\delta \sim 1$ Equation is solved numerically in few minutes





$$\dot{M} = rac{\pi a^2 P}{v_m} \left( -G_P \, rac{a}{P} \, rac{\mathrm{d}P}{\mathrm{d}x} + G_T \, rac{a}{T} \, rac{\mathrm{d}T}{\mathrm{d}x} 
ight)$$
  $G_P = G_P(\delta) \qquad G_T = G_T(\delta)$   $\delta = rac{Pa}{\mu v_m} \sim rac{1}{\mathsf{Kn}}$ 

Input equation to obtain  $G_{P_1}$ 

$$c_r \frac{\partial \phi}{\partial r} - \frac{c_\theta}{r} \frac{\partial \phi}{\partial \theta} = \delta(u - \phi) - \frac{1}{2}, \quad u = \frac{1}{\sqrt{\pi}} \int \mathbf{e}^{-c^2} \phi(x,c) \mathrm{d}c$$

Input equation to obtain  $G_P$ 

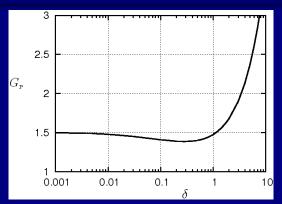
$$c_r \frac{\partial \phi}{\partial r} - \frac{c_\theta}{r} \frac{\partial \phi}{\partial \theta} = \delta(u - \phi) - \frac{1}{2}, \quad u = \frac{1}{\sqrt{\pi}} \int \mathrm{e}^{-c^2} \phi(x,c) \mathrm{d}c$$

Free-molecular regime,  $\delta = 0$  analytical solution

$$G_{\mathsf{P}} = \frac{8}{3\sqrt{\pi}}$$

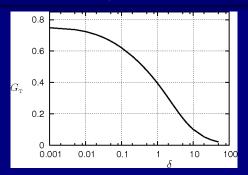
Transitional regime  $\delta \sim 1$ 

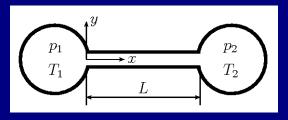
Equation is solved numerically in few minutes



Transitional regime  $\delta \sim 1$ 

Equation is solved numerically in few minutes

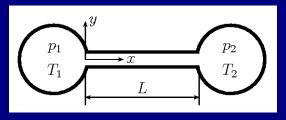




If  $p_1 \gg p_2$  and/or  $T_1 \gg T_2$  then Eq.

$$\dot{M} = \frac{\pi a^2 P}{v_m} \left( -G_P \frac{a}{P} \frac{\mathrm{d}P}{\mathrm{d}x} + G_T \frac{a}{T} \frac{\mathrm{d}T}{\mathrm{d}x} \right)$$

is integrated along  $\boldsymbol{x}$ 



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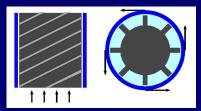
is integrated along x

Numerical calculations of  $\dot{M}$  can be carried out on-line http://fisica.ufpr.br/sharipov

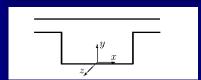


# Scheme of pump

#### Scheme of pump



#### Scheme of single groove



First stage

Four problems are solved for a single groove

#### First stage

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Longitudinal Poiseuille flow

#### First stage

Four problems are solved for a single groove

- Longitudinal Poiseuille flow
- Transversal Poiseuille flow

#### First stage

Four problems are solved for a single groove

- Longitudinal Poiseuille flow
- Transversal Poiseuille flow
- Longitudinal Couette flow

#### First stage

Four problems are solved for a single groove

- Longitudinal Poiseuille flow
- Transversal Poiseuille flow
- Longitudinal Couette flow
- Transversal Couette flow

Solution is determined by geometrical parameters of groove and by local rarefaction parameter.

This stage takes few days of computation.

Second stage stage

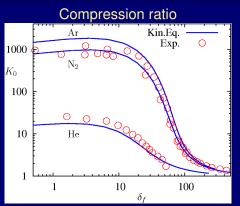
Compression ratio and pumping speed are calculated as a linear combinations of the four solutions.

Second stage stage

Compression ratio and pumping speed are calculated as a linear combinations of the four solutions.

This stage takes few seconds of computation.

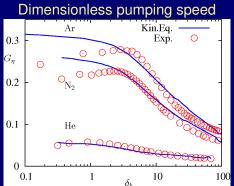
#### Comparison numerical and experimental results



Sharipov, Fahrenbach, and Zipp, JVSTA, Vol. 23, P.1331 (2005).



Comparison numerical and experimental results



Sharipov, Fahrenbach and Zipp, JVSTA, Vol. 23 (1331).

#### Advantages

No statistical noise.

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- Possibility to apply already obtained results

Disadvantages

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 Grids in both physical and velocity spaces must be carefully chosen

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- Difficult generalization for gaseous mixtures and polyatomic gases

Flows with high Mach number

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- Complicated geometrical configurations
- Flows with dissociation, recombinations, ionization etc.

Flows with low Mach number

- Flows with low Mach number
- Extended region of gas flows

- Flows with low Mach number
- Extended region of gas flows
- Simple geometrical configurations

- Flows with low Mach number
- Extended region of gas flows
- Simple geometrical configurations
- Flows without dissociation, recombinations, ionization etc.

## Thank you for your attention

http://fisica.ufpr.br/sharipov/