



DCA313 Shell Gauge Data During Welding Operation

Attached are shell strain gauge data taken during the welding of DCA313. The gauges are at the same locations as on DCA312 and the same analysis and plots are shown here as for 312[1]. The data are displayed in Table I. The gauge at 36 degrees near the lead end failed when the press was closed and it is not included in the table. The compensating gauge towards the return end (C2) failed during the welding process; its data are included in the table but not in the plots.

Figure 1 plots the data sequentially and Figure 2 shows the strain changes versus press load before welding. There were apparently no data taken after the upper press platen was lowered onto the magnet but before the hydraulic cylinders were energized. Otherwise the data follow the same pattern as DCA312: bending effects drive some gauges positive and others negative as the press closes, strains increase as the shell is welded, and then redistribute themselves slightly as the press is opened. Figure 3 shows the strain change from the press closure versus the angle from the weld. The pattern of bending effects is qualitatively the same as in DCA312. Figure 4 is a plot of the strain change due to welding. As in DCA312, there is a large change in the gauges at 17, 28 and 36 degrees, relatively little change at 61 degrees, and a somewhat larger change at 90 degrees. The change when the press is opened is plotted in Figure 5. As in DCA312 the stress redistributes slightly from the shell parting plane towards 90 degrees.

The effect of welding is a combination of azimuthal tension and further bending as the shell is pulled into better conformance with the yoke. Figure 6 shows crude attempts at correcting for the bending by subtracting from the raw strain change a multiple of the bending observed as the press is closed. Plotted there are the average of the two gauges versus angle from the weld, and this average corrected by subtracting 1 and 1.5 times the change with press closure. The case of a 1.5 times the press closure change yields a smooth distribution over the 17-36 degree gauges (which does not prove that this correction is correct). As with DCA312 the apparent strain change due to shell tension near the weld is on the order of 1500 microstrain, corresponding to a stress of 45 kpsi.

However, the corrected strain changes at 60 and 90 degrees are still quite small and, as with DCA312, a smooth distribution cannot be obtained by this simple correction procedure. It is quite possible that there really is a large loss of stress between the 36 and 61 degree measurements because the shell could be binding on the yoke pack pick-up features near 45 degrees[2]. This does not, however, explain how there can be a substantially larger strain change at 90 degrees than at 61 degrees. The gauges have already been mounted

on the shell for DCA314, but it is planned to change the locations of the gauges on DCA315 to try to understand this pattern.

REFERENCES

- [1] J. Strait, DCA312 Shell Gauge Data During Welding Operation, TS-SSC 91-179, 9/13/91.
- [2] This was pointed out to me by Eric Haggard.

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Table I

Weld - '

Weld -

DCA313 50mm Long Magnet Pre-weld Instrumentation Data Files R. Kunzelman & R. Tuskey T/S Engineering Lab CALCULATED STRAIN READINGS

Strain change due to welding

1881

1973

1927

1722

1619

-48

-5

-27

Strain change due to press opening Angle 17 28 36

17

28

2542

2332

2437

1845

1548

-58

-16

-37

36

1269

1269

1369

1419

Ø

-11

-6

61

15Ø

257

204

159

136

61

176

107

142

9Ø 599 577

588

415

329

9Ø

155

97

126

Angle Lead End

turn End

Average

-Closure

*Closure

Angle Lead End

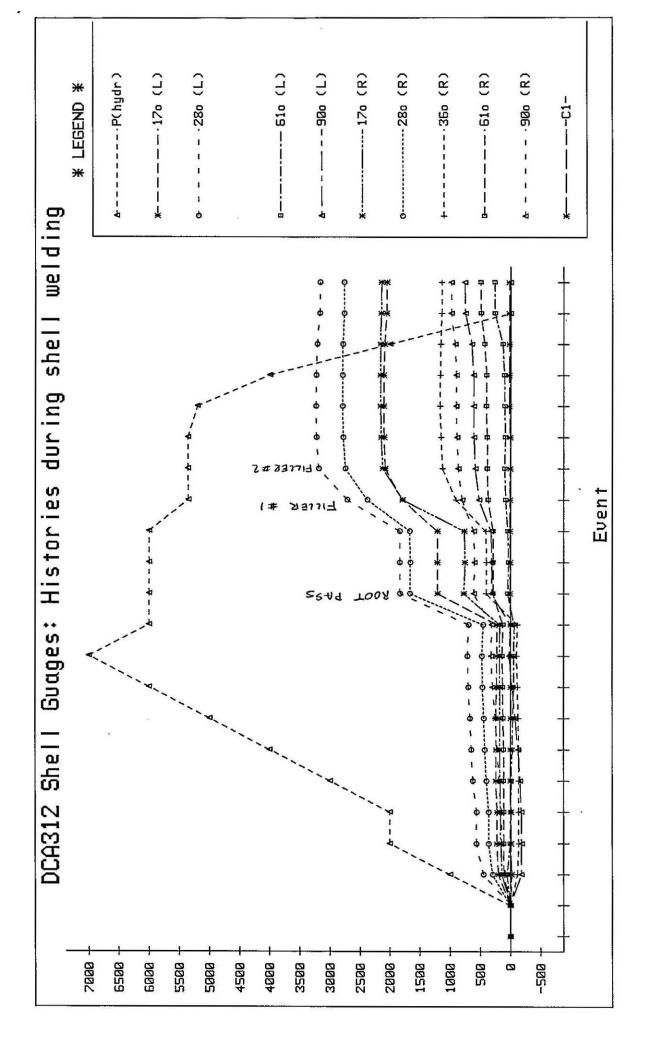
turn End

Average

Force	-A1-	-A2-	-A3-	-A4 -	-A5-	-A6-	-A7-	-A8-	-A9-	-A1Ø-	-C1-	-C2-					
P(hydr)	17o (L)	28o (L)		610 (L)		17o (R)	28o (R)	36o (R)		9Øo (R)	-C1-	-C2-			· L · L · L · C · C · C · C · C · C · C · C · C		
Ø	Ø	10		ø	Ø	Ø	ø	Ø		Ø	Ø	Ø		9/1	6/91	11:11:31	
0	2	3		2	2	2	200	2		200	2	2					
1000 2000	200 224	445 561		55 1Ø	-18Ø -182	141 16Ø	295			82	-20	-3					
2000	223	555		-1	-178	165	357 362	-139 -129		149 154	-11 -6	ø 1ø		0/1	0 /01		
3000	243	622		-14	-158	173	397	-12 <i>5</i> -137	114	189	2	10		9/1	6/91	13:07:43	
4000	237	648		-25	-126	179	424	-137	114	216	2	0					
5000	243	67Ø		-35	-78	179	440	-126	125	253	-3	0					
6000	234	693		-5ø	-23	180	463	-114	132	293	-1	9					
7000	227	713		-51	23	184	472	-100	141	323	2	13					
6000	223	688		-6Ø	4	18Ø	452	-114	127	3Ø9	-,1	9					
6000	1209	1833		43	313	768	1668	404	289	6Ø3	11	3	root	1e+	root	0/10/01	7.50.41
6000	1213	1837		35	311	754	166Ø	398	288	595	- 9	-4			root	9/18/91 9/18/91	7:50:41
6000	1211	1835		45	321	764	167Ø	406	292		13	6			1000	3/10/31	7:53:27
5350	1792	2719		8Ø	517	1791	2381	9Ø5	368	800	12	1643	fill	1st	fille	0/10/01	7:24:ØE
5360	2081	3191		96	586	2127	2746	1129		864	5	-20		2nd	fille	9/19/91	13:07:41
5361	2104	3230		90	603	2153	2784	1155	384	886	10	1676				3/13/31	13:07:41
5200	2111	3237		97	610	2160	2797	1172	401	904	22	1689					
4000	21Ø8	3234		104	612	2162	2798	1164	398	900	24	1691					
2000	2095	3216		129	643	2160	2792	1157	428	920	22	1694					
Ø	2Ø58	3174		258	749	2150	277Ø	1146	487	984	22	1689					
Ø	2056	3172		266	758	2148	2768	1144	491	983	26	1692					
Strain change due to press closure																	
Angle	17	28	36	61	9Ø												
Lead End	227	713		-51	23												
turn End	184	472	-100	141	323												
Average	206	593	-100	45	173												
-						• .											

1*closure

1.5*closure



Figurel

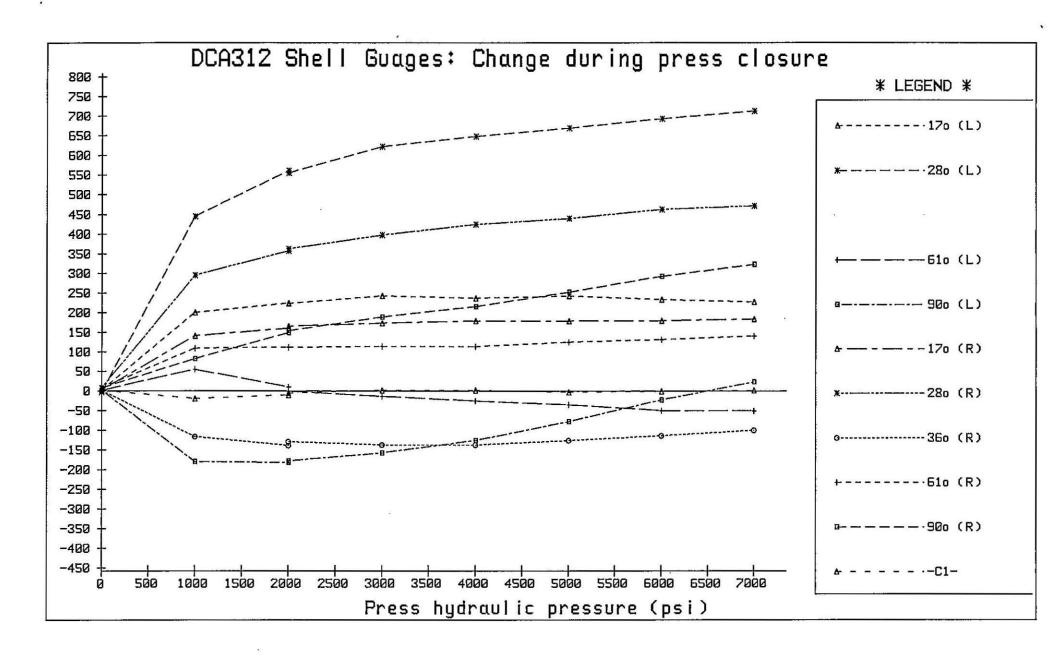


Figure 2

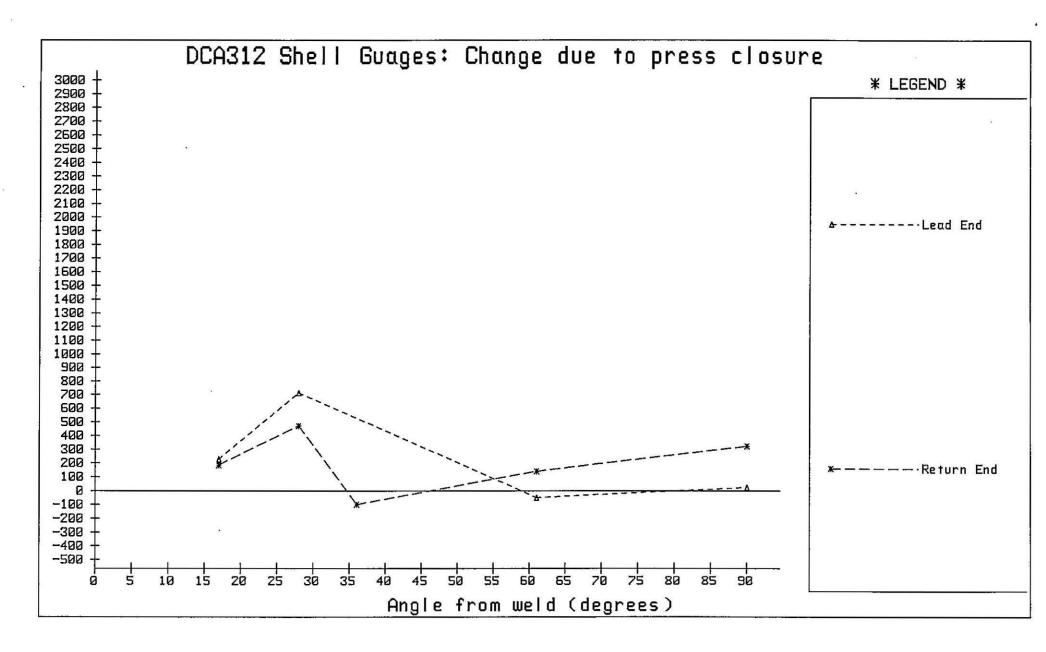


Figure 3

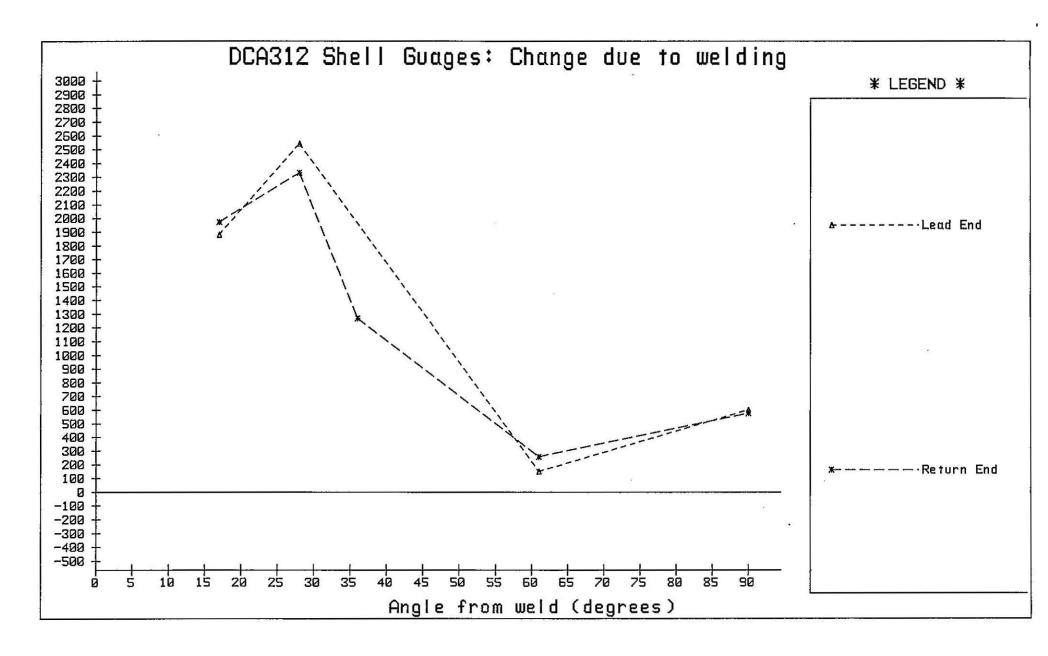


Figure 4

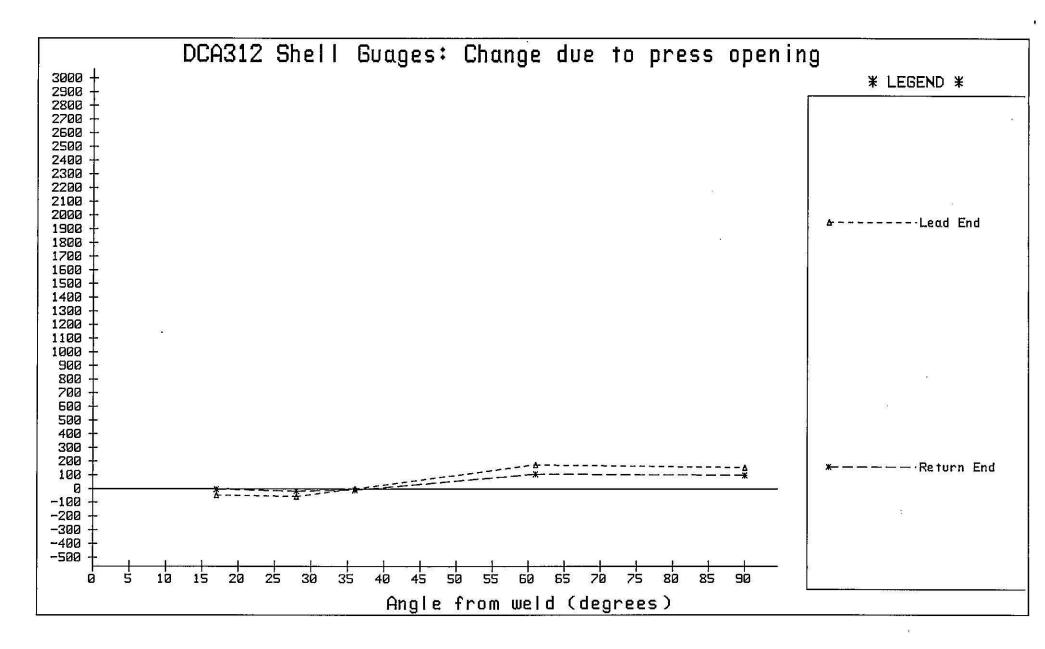


Figure 5

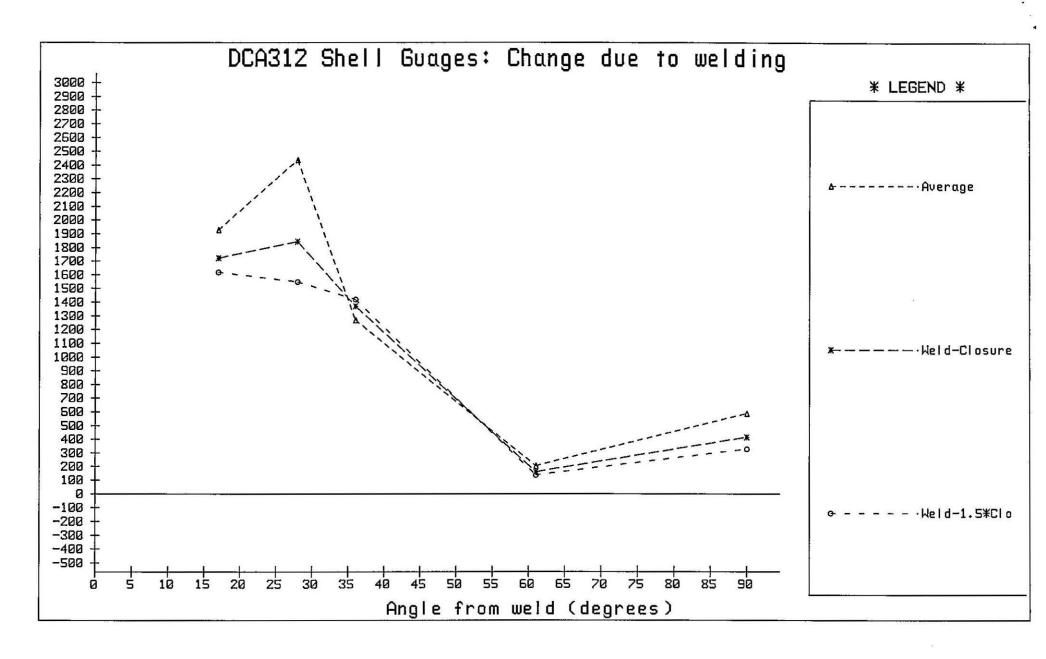


Figure 6