



**Fermi National Accelerator Laboratory**

**FERMILAB-Pub-92/231**

# **Covariance Matrices for Track Fitting with the Kalman Filter**

E.J. Wolin and L.L. Ho

*J.W. Gibbs Laboratory, Yale University  
New Haven, CT. 06511*

September 1992

Submitted to *Nuclear Instruments and Methods*  
Calculations for this paper were done for E791

## **Disclaimer**

*This report was prepared as an account of work sponsored by an agency of the United States Government. Neither the United States Government nor any agency thereof, nor any of their employees, makes any warranty, express or implied, or assumes any legal liability or responsibility for the accuracy, completeness, or usefulness of any information, apparatus, product, or process disclosed, or represents that its use would not infringe privately owned rights. Reference herein to any specific commercial product, process, or service by trade name, trademark, manufacturer, or otherwise, does not necessarily constitute or imply its endorsement, recommendation, or favoring by the United States Government or any agency thereof. The views and opinions of authors expressed herein do not necessarily state or reflect those of the United States Government or any agency thereof.*

# Covariance Matrices for Track Fitting with the Kalman Filter

E.J. Wolin\* and L.L. Ho†  
J.W. Gibbs Laboratory  
Yale University  
New Haven, CT 06511

February 24, 1993

## Abstract

We present a simple and intuitive derivation of the track parameter covariance matrix due to multiple Coulomb scattering for use in track fitting with the Kalman filter. We derive all the covariance matrix elements for two experimentally relevant track parameterizations (i.e.  $x$  and  $y$  slopes and intercepts, and direction cosines and intercepts) in the presence of thin scatterers and absence of magnetic fields. We further comment on how to account for thick and/or continuous scattering centers.

## 1 Introduction

Track fitting (including multiple Coulomb scattering) with the Kalman filter technique[1][2] has recently proven to be a powerful alternative to the traditional Wiener or “global” method, as well as to the breakpoint and various other ad hoc methods[3]–[12].

The Kalman filter is a recursive or stepwise procedure for estimating the state parameters of a linear, discrete dynamic system (e.g. a track in a particle physics detector). The Kalman method was originally introduced in 1960[1], and subsequently led to a revolution in various engineering fields, including

---

\*Bitnet: WOLIN@YALEHEP

†Bitnet: HO@YALEHEP

optimal signal processing, navigation, spacecraft tracking etc[13]. It was not until Fruhwirth's 1987 paper[2], however, that the Kalman filter technique became generally known in particle physics. Prior to this Billoir[4] independently derived equations equivalent to the Kalman filter equations without realizing it. With Fruhwirth's paper, however, the full power of the Kalman technique became available, especially the smoother algorithm.

The Kalman filter solution reproduces the result of the global least squares fit, but avoids the  $N \times N$  matrix inversion required by the latter method (where  $N$  is the total number of measurements). This matrix is in general non-diagonal due to correlations (caused by multiple scattering) between measurements. The largest matrix inversion required by the Kalman technique is the smaller of  $m \times m$  (gain matrix formalism, see below) and  $p \times p$  (weighted means formalism), where  $m$  is the number of measurements per point, and  $p$  is the number of parameters to be estimated (i.e. dimension of the state vector).

In contrast to the Kalman filter, the breakpoint method[7] reproduces the results of the global fit through actual fitting of the multiple scattering angles at each scattering plane. Ad hoc methods often only approximate the result of the global fit, and are generally used simply to avoid the costly matrix inversion.

An essential ingredient in the Kalman technique is the covariance matrix of the track parameters due to multiple Coulomb scattering. Although some partial results are available in the current literature[3][12], results are inconsistent, and derivations are often absent. In the following we offer a concise and general method for evaluating the covariance matrices in the presence of thin scattering centers and absence of magnetic fields.

The organization of our discussion is as follows. In section 2 we review the Kalman filter technique, and present a simplified derivation of the filter equations. In section 3 we define our notation and coordinate systems, and derive all terms of the covariance matrix for two experimentally relevant track parameterizations. We then comment on how our formalism can be used with thick and/or continuous scattering centers. In section 4 we discuss recent results concerning the variance of the projected scattering angle due to multiple scattering. Section 5 contains a summary and conclusions.

## 2 The Kalman Filter

The Kalman technique focuses on a  $p \times 1$  "state vector" that contains the  $p$  state parameters to be estimated, and on a model that extrapolates the state vector from point to point. These points can either be real points in space or time, or can simply be dimensionless integers (e.g. the track number in vertex fitting).

In the following, we use  $\bar{x}_k$  to denote the  $p \times 1$  state vector containing the true state parameters at point  $k$ . The state vector extrapolation model in the linear case is

$$\bar{x}_k = F_{k-1}\bar{x}_{k-1} + w_{k-1},$$

where  $F_{k-1}$  is a matrix that extrapolates the state vector from point  $k-1$  to point  $k$ , and  $w_{k-1}$  represents “process noise” that corrupts the state vector (in track fitting, for example, the process noise is due to multiple scattering). The process noise is assumed to be unbiased and to have finite variance, and its covariance matrix is  $Q_k$ .

The main result of this paper, given in section 3, is the evaluation of  $Q_k$  for two experimentally relevant track parameterizations.

The components of the state vector  $\bar{x}_k$  are not measured directly. The actual  $m$  measurements  $m_k$  at point  $k$  are linear functions of the state vector  $\bar{x}_k$  such that,

$$m_k = H_k\bar{x}_k + \epsilon_k,$$

where  $m_k$  is a  $m \times 1$  vector,  $H_k$  is a  $m \times p$  matrix, and  $\epsilon_k$  represents measurement noise, or measurement errors. In analogy to the process noise, the measurement noise  $\epsilon_k$  is assumed to be unbiased and to have finite variance, and its covariance matrix is  $V_k$ .

Following Gelb[14], we define  $\bar{x}_k^{k-1}$  to be the best estimate of the true state vector  $\bar{x}_k$  using all measurements up to but *not* including the  $k$ th measurement, and  $\bar{x}_k^k$  to be the best estimate of the state vector *including* the  $k$ th measurement. In the language of Kalman filtering,  $\bar{x}_k^{k-1}$  is the *predicted estimate* of  $\bar{x}_k$ , and  $\bar{x}_k^k$  is the *filtered estimate* of  $\bar{x}_k$ . Further, for  $n > k$ ,  $\bar{x}_k^n$  is called the *smoothed estimate* of  $\bar{x}_k$ .

In the Kalman scheme  $\bar{x}_k^k$  is taken to be an arbitrary linear function of the extrapolated (or predicted) value of the state vector,  $\bar{x}_k^{k-1}$ , and the actual measurement  $m_k$  made at point  $k$ :

$$\bar{x}_k^k = K_k^1\bar{x}_k^{k-1} + K_k^2m_k,$$

where  $K_k^1$  and  $K_k^2$  are arbitrary matrices. The Kalman prediction equation is taken to be

$$\bar{x}_k^{k-1} = F_{k-1}\bar{x}_{k-1}^{k-1}.$$

Requiring  $\bar{x}_k^k$  to be unbiased (i.e. expectation value of  $(\bar{x}_k^k - \bar{x}_k) = 0$ ) yields

$$\bar{x}_k^k = \bar{x}_k^{k-1} + K_k(m_k - H_k\bar{x}_k^{k-1})$$

where  $K_k$  is called the Kalman gain matrix. Further,

$$K_k^1 = (I - K_kH_k)$$

and

$$K_k^2 = K_k.$$

$K_k$  is determined by requiring that it minimize the sum of the squares of the standard deviations of the estimated parameters,

$$\frac{\partial T\tau(C_k)}{\partial K_k} = 0,$$

where  $T\tau(C_k)$  is the trace of  $C_k$  and  $C_k$  is the  $p \times p$  covariance matrix of the estimated parameters at point  $k$ . Solving for  $K_k$  yields

$$K_k = C_k^{k-1} H_k^T (V_k + H_k C_k^{k-1} H_k^T)^{-1},$$

where

$$C_k^{k-1} = F_{k-1} C_{k-1}^{k-1} F_{k-1}^T + Q_{k-1}$$

and

$$C_k^k = (1 - K_k H_k) C_k^{k-1}$$

(the superscript  $T$  denotes the transpose). In correspondence to our earlier notation,  $C_k^{k-1}$  is the best estimate of  $C_k$  excluding the  $k^{\text{th}}$  measurement, and  $C_k^k$  is the best estimate of  $C_k$  including the  $k^{\text{th}}$  measurement.

We note that our equations correspond to the ‘‘gain matrix’’ formulation of the Kalman filter method. Fruhwirth[2] further explores the ‘‘weighted means’’ formulation, which focuses on the  $K_k^1$  and  $K_k^2$  matrices instead. The two methods are mathematically equivalent; the gain matrix formulation is generally preferred if  $m < p$ .

Full discussions of the Kalman filter, Kalman smoother, and other approaches to track fitting including multiple scattering can be found in references [2]–[21].

## 3 Calculation of Covariance Matrices

### 3.1 Notation and Outline of Calculation

Multiple scattering is conveniently parameterized by two mutually orthogonal, uncorrelated scattering angles  $\theta_1$  and  $\theta_2$ [3][12][22], which are assumed to be small. We ignore the transverse displacement of the track as it crosses the scattering plane[23], and assume that the scattering takes place at a single point in the center of the scattering plane.

The covariance matrix elements  $\langle \theta_i, \theta_j \rangle$  for the scattering angles  $\theta_1$  and  $\theta_2$  can be written as

$$\langle \theta_i, \theta_j \rangle = \sigma^2(\theta_{proj}) \delta_{ij}, \quad (1)$$

where  $\sigma^2(\theta_{proj})$  is the variance of the projected multiple scattering angle (see section 4 for an explicit form for  $\sigma(\theta_{proj})$ ).

The covariance matrix elements  $\langle P_i, P_j \rangle$  for any two arbitrary functions  $P_i(\theta_1, \theta_2)$  and  $P_j(\theta_1, \theta_2)$  can be evaluated by using the propagation of errors formula[22]

$$\langle P_i, P_j \rangle = \sum_{m,n=1}^2 \frac{\partial P_i}{\partial \theta_m} \frac{\partial P_j}{\partial \theta_n} \langle \theta_m, \theta_n \rangle. \quad (2)$$

In the present context,  $\langle P_i, P_j \rangle$  represents the elements of the covariance matrix  $Q_k$ , defined in section 2. Substitution of Eq. 1 into equation Eq. 2 yields

$$\langle P_i, P_j \rangle = \sigma^2(\theta_{proj}) \left( \frac{\partial P_i}{\partial \theta_1} \frac{\partial P_j}{\partial \theta_1} + \frac{\partial P_i}{\partial \theta_2} \frac{\partial P_j}{\partial \theta_2} \right). \quad (3)$$

Hence, calculation of the covariance matrix elements for any track parameterization requires evaluation of the partial derivatives of the track parameters with respect to the two orthogonal scattering angles (in the limit  $\theta_1, \theta_2 \rightarrow 0$ ).

The outline of our derivation is as follows. We express the unscattered track parameters in a *reference* coordinate system, let the track to scatter at the point  $(x_0, y_0, z_0)$ , and recalculate the scattered track parameters in the reference system. The scattering angles  $\theta_1$  and  $\theta_2$  are defined in a *track* coordinate system whose origin is  $(x_0, y_0, z_0)$ , and whose  $z$  axis is aligned with the unscattered track (note that although the  $z$  axis of the track system is aligned with the unscattered track, the orientation of the other two axes is arbitrary). We then calculate the partial derivatives of the scattered track parameters with respect to  $\theta_1$  and  $\theta_2$  in the limit  $\theta_1, \theta_2 \rightarrow 0$ , and use the propagation of errors formula to obtain our final result.

The angles between the projections of the unscattered (scattered) track onto the  $x$ - $z$  and  $y$ - $z$  planes of the reference system and the  $z$ -axis of the reference system are  $\theta_x$  and  $\theta_y$  ( $\theta_x^*$  and  $\theta_y^*$ ) respectively. Similarly, the angles between the projection of the scattered track onto the  $x$ - $z$  and  $y$ - $z$  planes of the track system and the  $z$ -axis of the track system are  $\theta_1$  and  $\theta_2$ . In the limit  $\theta_1, \theta_2 \rightarrow 0$ ,  $\theta_x^*, \theta_y^* \rightarrow \theta_x, \theta_y$ .

Finally, the direction cosines of the  $x$ ,  $y$ , and  $z$  axes of the track system in the reference system are  $(\alpha_1, \beta_1, \gamma_1)$ ,  $(\alpha_2, \beta_2, \gamma_2)$ , and  $(\alpha_3, \beta_3, \gamma_3)$  respectively.

### 3.2 Direction Tangent Case

For the direction tangent case, we define the  $4 \times 1$  state vector  $\bar{x}$  as:

$$\bar{x} = \begin{pmatrix} P_1 \\ P_2 \\ P_3 \\ P_4 \end{pmatrix} = \begin{pmatrix} x_0 - z_0 \tan \theta_x \\ y_0 - z_0 \tan \theta_y \\ \tan \theta_x \\ \tan \theta_y \end{pmatrix}. \quad (4)$$

That is,  $(P_1, P_2, 0)$  is the intercept of the unscattered track with the reference x-y plane at  $z = 0$ , and  $P_3$  and  $P_4$  are the direction tangents (or slopes,  $dx/dz$  and  $dy/dz$ ) of the unscattered track in the reference system. In the track coordinate system the incoming (unscattered) track unit vector is simply

$$\begin{pmatrix} 0 \\ 0 \\ 1 \end{pmatrix},$$

while after scattering the unit vector becomes

$$\frac{1}{\sqrt{1 + \tan^2 \theta_1 + \tan^2 \theta_2}} \begin{pmatrix} \tan \theta_1 \\ \tan \theta_2 \\ 1 \end{pmatrix}. \quad (5)$$

The  $3 \times 3$  rotation matrix connecting the track and reference systems is

$$R = \begin{pmatrix} \alpha_1 & \alpha_2 & \alpha_3 \\ \beta_1 & \beta_2 & \beta_3 \\ \gamma_1 & \gamma_2 & \gamma_3 \end{pmatrix},$$

with elements obeying the relations

$$\begin{aligned} \alpha_i \alpha_j + \beta_i \beta_j + \gamma_i \gamma_j &= \delta_{ij}, \\ \alpha_1 \beta_1 + \alpha_2 \beta_2 + \alpha_3 \beta_3 &= 0, \\ \alpha_1 \gamma_1 + \alpha_2 \gamma_2 + \alpha_3 \gamma_3 &= 0, \\ \beta_1 \gamma_1 + \beta_2 \gamma_2 + \beta_3 \gamma_3 &= 0. \end{aligned} \quad (6)$$

Thus the scattered unit vector in the reference system is

$$a \begin{pmatrix} \tan \theta_x^* \\ \tan \theta_y^* \\ 1 \end{pmatrix} = a' \begin{pmatrix} \alpha_1 & \alpha_2 & \alpha_3 \\ \beta_1 & \beta_2 & \beta_3 \\ \gamma_1 & \gamma_2 & \gamma_3 \end{pmatrix} \begin{pmatrix} \tan \theta_1 \\ \tan \theta_2 \\ 1 \end{pmatrix}, \quad (7)$$

where  $a$  and  $a'$  are normalization constants for the unit vectors. Solving Eq. 7 for the direction tangents of the scattered vector in the reference system yields

$$P_3 = \tan \theta_x^* = \frac{\alpha_1 \tan \theta_1 + \alpha_2 \tan \theta_2 + \alpha_3}{\gamma_1 \tan \theta_1 + \gamma_2 \tan \theta_2 + \gamma_3}, \quad (8)$$

$$P_4 = \tan \theta_y^* = \frac{\beta_1 \tan \theta_1 + \beta_2 \tan \theta_2 + \beta_3}{\gamma_1 \tan \theta_1 + \gamma_2 \tan \theta_2 + \gamma_3}. \quad (9)$$

Taking the partial derivatives of  $P_3$  and  $P_4$  with respect to  $\theta_1$  and  $\theta_2$  in the small angle limit gives

$$\left. \frac{\partial P_3}{\partial \theta_1} \right|_{\theta_1, \theta_2 \rightarrow 0} = \frac{\alpha_1 \gamma_3 - \alpha_3 \gamma_1}{(\gamma_3)^2}, \quad (10)$$

$$\left. \frac{\partial P_3}{\partial \theta_2} \right|_{\theta_1, \theta_2 \rightarrow 0} = \frac{\alpha_2 \gamma_3 - \alpha_3 \gamma_2}{(\gamma_3)^2}, \quad (11)$$

$$\left. \frac{\partial P_4}{\partial \theta_1} \right|_{\theta_1, \theta_2 \rightarrow 0} = \frac{\beta_1 \gamma_3 - \beta_3 \gamma_1}{(\gamma_3)^2}, \quad (12)$$

$$\left. \frac{\partial P_4}{\partial \theta_2} \right|_{\theta_1, \theta_2 \rightarrow 0} = \frac{\beta_2 \gamma_3 - \beta_3 \gamma_2}{(\gamma_3)^2}. \quad (13)$$

Furthermore,

$$\begin{aligned} \frac{\partial P_1}{\partial \theta_1} &= -z_0 \frac{\partial P_3}{\partial \theta_1}, \\ \frac{\partial P_1}{\partial \theta_2} &= -z_0 \frac{\partial P_3}{\partial \theta_2}, \\ \frac{\partial P_2}{\partial \theta_1} &= -z_0 \frac{\partial P_4}{\partial \theta_1}, \\ \frac{\partial P_2}{\partial \theta_2} &= -z_0 \frac{\partial P_4}{\partial \theta_2}. \end{aligned}$$

Substituting the above partial derivatives into Eq. 3, and simplifying with Eqs. 6 yields

$$\begin{aligned} \langle P_3, P_3 \rangle &= \sigma^2(\theta_{proj}) \left\{ \left[ \frac{\alpha_1 \gamma_3 - \alpha_3 \gamma_1}{(\gamma_3)^2} \right]^2 + \left[ \frac{\alpha_2 \gamma_3 - \alpha_3 \gamma_2}{(\gamma_3)^2} \right]^2 \right\} \\ &= \sigma^2(\theta_{proj}) \left[ \frac{(\alpha_3)^2 + (\gamma_3)^2}{(\gamma_3)^4} \right]. \end{aligned} \quad (14)$$

Note that this result only depends on  $(\alpha_3, \beta_3, \gamma_3)$ , i.e. on the orientation of the  $z$  axis of the track system with respect to the reference system. Using the identities

$$\begin{aligned} \alpha_3 &= \frac{\tan \theta_x}{(1 + \tan^2 \theta_x + \tan^2 \theta_y)^{1/2}}, \\ \beta_3 &= \frac{\tan \theta_y}{(1 + \tan^2 \theta_x + \tan^2 \theta_y)^{1/2}}, \\ \gamma_3 &= \frac{1}{(1 + \tan^2 \theta_x + \tan^2 \theta_y)^{1/2}}, \end{aligned} \quad (15)$$

$\langle P_3, P_3 \rangle$  becomes (in the limit  $\theta_1, \theta_2 \rightarrow 0$ ,  $\theta_x^*, \theta_y^* \rightarrow \theta_x, \theta_y$ )

$$\begin{aligned} \langle P_3, P_3 \rangle &= \sigma^2(\theta_{proj}) (1 + \tan^2 \theta_x + \tan^2 \theta_y)(1 + \tan^2 \theta_x) \\ &= \sigma^2(\theta_{proj}) [1 + (P_3)^2][1 + (P_3)^2 + (P_4)^2]. \end{aligned} \quad (16)$$

Similarly,

$$\langle P_4, P_4 \rangle = \sigma^2(\theta_{proj}) [1 + (P_4)^2][1 + (P_3)^2 + (P_4)^2]. \quad (17)$$

In the same manner,

$$\begin{aligned}\langle P_3, P_4 \rangle &= \sigma^2(\theta_{proj}) \frac{\alpha_3 \beta_3}{(\gamma_3)^4} \\ &= \sigma^2(\theta_{proj}) P_3 P_4 [1 + (P_3)^2 + (P_4)^2].\end{aligned}\quad (18)$$

Finally, the full covariance matrix of the track parameters, constructed from the preceding three elements (i.e. Eqs. 16, 17 and 18), is

$$\begin{pmatrix} z_0^2 \langle P_3, P_3 \rangle & z_0^2 \langle P_3, P_4 \rangle & -z_0 \langle P_3, P_3 \rangle & -z_0 \langle P_3, P_4 \rangle \\ z_0^2 \langle P_3, P_4 \rangle & z_0^2 \langle P_4, P_4 \rangle & -z_0 \langle P_3, P_4 \rangle & -z_0 \langle P_4, P_4 \rangle \\ -z_0 \langle P_3, P_3 \rangle & -z_0 \langle P_3, P_4 \rangle & \langle P_3, P_3 \rangle & \langle P_3, P_4 \rangle \\ -z_0 \langle P_3, P_4 \rangle & -z_0 \langle P_4, P_4 \rangle & \langle P_3, P_4 \rangle & \langle P_4, P_4 \rangle \end{pmatrix}.$$

### 3.3 Direction Cosine Case

For the direction cosine case the  $4 \times 1$  state vector  $\bar{x}$  is

$$\bar{x} = \begin{pmatrix} P_1 \\ P_2 \\ P_3 \\ P_4 \end{pmatrix} = \begin{pmatrix} x_0 - z_0(\alpha_3/\gamma_3) \\ y_0 - z_0(\beta_3/\gamma_3) \\ \alpha_3 \\ \beta_3 \end{pmatrix},$$

where  $\alpha_3$ ,  $\beta_3$ , and  $\gamma_3$  are the direction cosines of the incoming (unscattered) track in the reference system. Proceeding as before, and using the definition of direction cosines, we have

$$\begin{pmatrix} \alpha^* \\ \beta^* \\ \gamma^* \end{pmatrix} = a \begin{pmatrix} \alpha_1 & \alpha_2 & \alpha_3 \\ \beta_1 & \beta_2 & \beta_3 \\ \gamma_1 & \gamma_2 & \gamma_3 \end{pmatrix} \begin{pmatrix} \tan \theta_1 \\ \tan \theta_2 \\ 1 \end{pmatrix},$$

where  $a$  is a normalization constant, and  $\alpha^*$ ,  $\beta^*$ , and  $\gamma^*$  are the direction cosines of the scattered track in the reference system (note that in the limit  $\theta_1, \theta_2 \rightarrow 0$ ,  $(\alpha^*, \beta^*, \gamma^*) \rightarrow (\alpha_3, \beta_3, \gamma_3)$ ). Evaluating the partial derivatives of the scattered track parameters in the limit  $\theta_1, \theta_2 \rightarrow 0$  gives

$$\begin{aligned}\partial \alpha^* / \partial \theta_1 &= \alpha_1, & \partial \alpha^* / \partial \theta_2 &= \alpha_2, \\ \partial \beta^* / \partial \theta_1 &= \beta_1, & \partial \beta^* / \partial \theta_2 &= \beta_2, \\ \partial \gamma^* / \partial \theta_1 &= \gamma_1, & \partial \gamma^* / \partial \theta_2 &= \gamma_2.\end{aligned}$$

Finally, using the propagation of errors formula and the orthogonality relations of the rotation matrix, the full covariance matrix of the track parameters

is

$$\sigma^2(\theta_{proj}) \times \begin{pmatrix} \left[ \frac{z_0^2(1-P_4^2)}{(1-P_3^2-P_4^2)^2} \right] & \left[ \frac{z_0^2 P_3 P_4}{(1-P_3^2-P_4^2)^2} \right] & \left[ \frac{-z_0}{(1-P_3^2-P_4^2)^{1/2}} \right] & 0 \\ \left[ \frac{z_0^2 P_3 P_4}{(1-P_3^2-P_4^2)^2} \right] & \left[ \frac{z_0^2(1-P_3^2)}{(1-P_3^2-P_4^2)^2} \right] & 0 & \left[ \frac{-z_0}{(1-P_3^2-P_4^2)^{1/2}} \right] \\ \left[ \frac{-z_0}{(1-P_3^2-P_4^2)^{1/2}} \right] & 0 & 1 - P_3^2 & -P_3 P_4 \\ 0 & \left[ \frac{-z_0}{(1-P_3^2-P_4^2)^{1/2}} \right] & -P_3 P_4 & 1 - P_4^2 \end{pmatrix}.$$

### 3.4 Comments

If the transverse displacement of the track as it crosses the scattering plane cannot be ignored, two modifications must be made. First, the error matrix for the multiple scattering angles  $\theta_1$  and  $\theta_2$  (Eq. 1) must be replaced by the full  $4 \times 4$  correlated error matrix of multiple scattering angles and displacements[3][12]. Second, the partial derivatives of the track parameters with respect to the two transverse displacements must be calculated, following the method given in the preceding sections; Eq. 2 (rather than Eq. 3) is then used.

If the scattering centers are too thick to be considered discrete, the formula for the mean scattering angles (see section 4) and displacements[22] must be replaced by an integral over the scattering medium[3][12].

## 4 Variance of Multiple Scattering Angle

Several forms of the standard deviation  $\sigma(\theta_{proj})$  are available. Lynch and Dahl[22][24] obtain, with result good to approximately 11%,

$$\sigma(\theta_{proj}) = \frac{13.6}{\beta c p} z \sqrt{L/L_r} [1 + 0.038 \ln(L/L_r)], \quad (19)$$

where  $L$  is the path length through the material,  $L_r$  is the radiation length of the material,  $\beta c$  is the velocity of the particle,  $z$  is the charge, and  $p$  is the momentum in MeV. This result is a modification of the previous formulae of Rossi-Griesen[26] and Highland[27] (who introduced the logarithmic term).

Lynch and Dahl also present an alternative form, good to better than 2%, that uses Moliere's characteristic angle and screening angle[25] instead of the radiation length.

## 5 Conclusions

The application of the Kalman filter technique to track fitting requires evaluation of the covariance matrix of the track parameters due to multiple scattering. Results in the current literature are generally incomplete, and derivations are often absent.

We have briefly introduced the basics of Kalman filter technique, and have calculated the full covariance matrix for two experimentally relevant track parameterizations for thin scatterers in the absence of a magnetic field: direction tangents and intercepts, and direction cosines and intercepts. Where they overlap, our results agree with the partial ones quoted in reference [12].

## Acknowledgment

This work is supported by the United States Department of Energy under grant number DE-AC02-76ER-03075.

## References

- [1] R.E. Kalman, *Trans. ASME, J. Bas. Eng.* **82D**, 35 (1960), and R.E. Kalman and R.S. Bucy, *ASME, J. Bas. Eng.* **83D**, 95 (1961).
- [2] R. Fruhwirth, *Nucl. Instr. and Meth.* **A262**, 444 (1987), and *Application of Filter Methods to the Reconstruction of Tracks and Vertices in Events of Experimental High Energy Physics*, Ph.D thesis, HEPHY-PUB 516/88, Institut fur Hochenergiephysik der Osterreichischen Akademie der Wissenschaften, A-1040 Wien, Nikolsdorfergasse 18, Vienna, 1988.
- [3] H. Eichinger and M. Regler, CERN 81-06 (1981).
- [4] P. Billoir, *Nucl. Instr. and Meth.* **225**, 352 (1984), and *Comp. Phys. Comm.* **57**, 390 (1989).
- [5] P. Billoir, R. Fruhwirth, and M. Regler, *Nucl. Instr. and Meth.* **A241**, 115 (1985).
- [6] P. Billoir and S. Qian, *Nucl. Instr. and Meth.* **A294** 219 (1990), and **A295** 492 (1990).
- [7] G. Lutz, *Nucl. Instr. and Meth.* **A273**, 349 (1988).

- [8] R.L. Gluckstern, Nucl. Instr. and Meth. **24**, 381 (1963).
- [9] J.C. Hart and D.H. Saxon, Nucl. Instr. and Meth. **220** 309 (1984).
- [10] D.H. Saxon, Nucl. Instr. and Meth. **A234** 258 (1985).
- [11] M. Regler and R. Fruhwirth, Reconstruction of Charged Tracks, in *Proceedings of the Advanced Study Institute on Techniques and Concepts in High Energy Physics*, Plenum Publ. Corp, 1988.
- [12] *Data Analysis Techniques for High-Energy Physics Experiments*, R. Bock, H. Grote, D. Notz, and M. Regler, Cambridge University Press, 1990.
- [13] *Kalman Filtering: Theory and Application*, H.W. Sorenson ed., IEEE Press, 1985. Included are popular accounts, original papers, technical analyses of the Kalman filter method, and numerous articles on the impact of Kalman filtering in engineering and related fields.
- [14] *Applied Optimal Estimation*, A. Gelb, MIT Press, 1975.
- [15] *Introduction to Random Signal Analysis and Kalman Filtering*, R. Brown, Wiley and Sons, 1983.
- [16] *Introduction to Kalman Filtering with Applications*, K. Miller and D. Leskiw, Krieger Publications, 1987.
- [17] *Stochastic Processes and Filtering Theory*, A. Jazwinski, Academic Press, 1970.
- [18] *Digital and Kalman Filtering*, S. Bozic, E. Arnold, 1979.
- [19] *Kalman-Bucy-Filter*, K. Brammer and G. Siffing, R. Oldenbourg, 1975.
- [20] D. Savard, B. Lorazo, and H. Jeremie, Nucl. Instr. and Meth. **A268** 278 (1988).
- [21] L. Stanco, Comp. Phys. Comm. **57** 380 (1989).
- [22] Particle Data Group, "Review of Particle Properties", Phys. Lett. B **239**, (1990)
- [23] For a 300  $\mu\text{m}$  thick silicon plane, the mean transverse displacement of a 1 gev track is approximately 0.1  $\mu\text{m}$ , much smaller than the typical readout pitch of 25 or 50  $\mu\text{m}$ .
- [24] G. Lynch and O. Dahl, Nucl. Instr. and Meth. **B58**, 6 (1991). This paper has an important typographical error. The left hand side of Eq. 7 should be  $\sigma^2$  rather than  $\sigma$ .
- [25] H.A. Bethe, Phys. Rev **89**, 1256 (1953).
- [26] B. Rossi and K. Greisen, Rev. Mod. Phys **13**, 240 (1941); and *High Energy Particles*, B. Rossi, Prentice-Hall, 1961.
- [27] V. Highland, Nucl. Instr. and Meth. **129**, 497 (1975), and Nucl. Instr. and Meth. **161**, 171 (1979).